## **Numerical Project**

- 1. Read handout "The restricted 3-body problem (2D)".
- 2. Run the Matlab code "restricted2d.m" with the subroutine "fnewton.m". First, set the input parameters to the following values (these are the defaults):

```
massratio = 0;
x20
      = 1;
v2x0
      = 0;
y20
      = 0;
v2y0 = sqrt(1/x20);
x30
      = 0;
y30
      = 2;
v3x0
      = -sqrt(1/y30)*0.5;
     = 0;
v3y0
      = 1*period2;
tmax
```

Answer the following questions:

- (a) What kind of orbit is  $m_2$  in?
- (b) What is its orbital period (in terms of T, the natural time scale)?
- (c) If you were to change the value of x20 would  $m_2$  still be in a circular orbit? Why or why not? (Try it! You might want to adjust tmax so that you follow the orbits longer.)
- (d) What does massratio = 0 mean physically?
- (e) What kind of orbit is  $m_3$  in?
- (f) Calculate its period (in terms of the period of  $m_2$ ).
- 3. Increase the time span to tmax = 10\*period2 and increase the effect of  $m_2$  to massratio = 0.01.

Answer the following questions:

- (a) What kind of orbit is  $m_3$  in now?
- (b) Take the final (vector) value of  $\mathbf{x}$ , which I have saved in the variable  $\mathbf{xnew}$ , run the simulation again with this as the initial value  $\mathbf{x0}$  and change  $\mathbf{massratio}$  back to zero. Determine the period and semimajor axis of  $m_3$  in this final state. Do they satisfy Kepler's third law?

- 4. Increase the perturbing effect of  $m_2$  to massratio = 0.1. Notice the effect of the close approaches of the two "planets". (NOTE: Don't forget to comment out x0 = xnew;)

  Answer the following questions:
  - (a) What happens to  $m_3$ ?
  - (b) Change the initial conditions very slightly, let y30 = 2.01. Now what happens to  $m_3$ ? Explain the results.
  - (c) Take the final (vector) value of  $\mathbf{x}$ , run the simulation again with this as the initial value  $\mathbf{x0}$  and change massratio back to zero. Determine the period and semimajor axis of  $m_3$  in this final state. Do they satisfy Kepler's third law?
- 5. Simulate the Earth and the Moon. (NOTE: Don't forget to comment out x0 = xnew;) Let  $m_2$  be the Earth and  $m_3$  be the moon.
  - (i) Calculate the necessary value for massratio.
  - (ii) Determine the initial conditions for  $m_3$ . One method is to use the defaults for  $m_2$ , but the following for  $m_3$

```
x30 = x20 + emdistance;
y30 = 0;
v3x0 = 0;
v3y0 = v2y0 + deltav;
```

where emdistance is the Earth-Moon distance (in the proper scaled units) and deltav is the speed of the Moon relative to the Earth (again, in proper dimensionless units).

Answer the following questions:

- (a) What is the proper value for massratio?
- (b) What are the proper values of emdistance and deltay?
- (c) Prove that the Moon has a stable orbit around the Earth by extending the simulation to tmax = 50\*period2, i.e., 50 "years."
  - i. You may find it useful to use figure(3) since it plots the moon centered on the Earth.
  - ii. By "stable" I mean that the moon remains in orbit and does not escape. You may notice, however, that it appears to lose energy. The reason is that energy is not conserved in the restricted three-body problem, but another quantity is. See § 2.12.2 on the Jacobi constant in Curtis's text for an explanation.
- (d) Reduce deltav by 50% so that the satellite (no longer the Moon) is in an elliptical orbit. Does the line of apsides retain a constant orientation? Explain why or why not.

6. Investigate some interesting scenario regarding  $m_2$  and  $m_3$ .

This last part of the project is open-ended. You are to play around with the initial conditions of  $m_3$  and see what happens. Try to obtain some interesting effects, such as a gravity assist that ejects  $m_3$  from the solar system, or a gravity assist that sends  $m_3$  onto an elliptical orbit.

Or you could look at the resonance between  $m_2$  and  $m_3$ . Set deltav to zero, and try several different small values for emdistance. Describe what you find.

Type up your answers to all of the above questions, along with commentary explaining the results. Be sure to include orbital plots, since "a picture is worth a thousand words," but your plots *must* have properly labeled axes and enough notation for the reader (me) to understand.